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a central station connected to said communication network configured to
receive and analyze said transmission of digital performance data
wherein said digital data includes an identifier unique to a particular
aircraft

- 5. The aircraft maintenance system of claim 4 wherein said transmitter is positionable on an aircraft having a flight data recorder and at least a portion of said digital performance data comprises data directed to said flight data recorder.
  - 6. The aircraft maintenance system of claim 4 further comprising:

    a sensor multiplexer located on said aircraft, said sensor

    multiplexer having a plurality of inputs for receiving aircraft performance

    and control parameters from existing aircraft sensors, and an output in

    communication with said transmitter for providing said digital

    performance data to said transmitter.
- 7. The aircraft maintenance system of claim 4 wherein said digital performance data further includes digitized audio information.

PATENT Attorney Docket No.: 57127 Preliminary Amendment Page 3 of 23

1	1458	8.	The aircraft maintenance system of claim 4 wherein said digital
2	perfo	rmance	data further includes digitized video information.
1		9.	The aircraft maintenance system of claim 5 wherein said digital
2 H C C C 1 L U 2 Q	perfo	rmance	data includes aircraft position data directed to said flight data recorder.
1 = 1		10.	The aircraft maintenance system of claim 9 wherein information provided
	by a	GPS rec	eiver is used in the calculation of said aircraft position data.
1 M 2 H	by an	11. inertial	The aircraft maintenance system of claim 10 wherein information provided navigation system is used in the calculation of said aircraft position data.
1		12.	The aircraft maintenance system of claim 4, wherein said central station is
2	furth	er config	gured to transmit digital data on said communication network, further
3	comp	rising:	
4			a receiver on said aircraft configured to receive digital data from said
5			communication network; and
6			a maintenance communication means, located on said aircraft, for

7	providing maintenance advice to maintenance personnel, said
8	maintenance communication means having an input for receiving
9	said maintenance advice from said receiver,
10	wherein said maintenance advice is transmitted from said central station to
11	said receiver.

13. The aircraft maintenance system of claim 12 wherein said maintenance advice is provides aurally to said maintenance personnel.

14. The aircraft maintenance system of claim 8 wherein said central station includes a storage system for storing said aircraft performance and control parameters.

not in claim but,

1	15.	An aircraft maintenance system comprising:
2		a transmitter configured for transmission of data across a communication
3		network, said transmitter positionable to be located on an aircraft;
4		a ground based station connected to said communication network
5		configured to receive said transmission of data; and
6		a sensor multiplexer located on said aircraft, said sensor multiplexer

having a plurality of inputs for receiving aircraft performance and

particular aircraft.

transmitter: wherein said digital data further includes an aircraft identifier unique to a

**PATENT** 

Page 5 of 23

Attorney Docket No.: 57127 Preliminary Amendment

16. The aircraft maintenance system of claim 15, wherein said ground based station is further configured to transmit data on said communication network, further comprising:

> a receiver located on said aircraft, said receiver configured to receive data from said communication network; and

> a maintenance communication means which receives maintenance advisory data from said receiver and provides maintenance advice to maintenance personnel,

wherein said maintenance advice is transmitted from said ground based station to said receiver.

stre data 17. The aircraft maintenance system of claim 15 wherein said ground based station includes a storage system for archiving said aircraft performance and control

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parameters.

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1	102	18.	A method for real-time monitoring and archiving of aircraft performance
2	102		data including the steps of:
3	5381 140		providing a performance sensor in an aircraft, said performance sensor
4 <del>  4</del>			having an output indicative of an aircraft performance parameter;
50			electronically transmitting at least said aircraft performance parameter to a
415 50 64 70			global communication network;
			receiving said aircraft performance parameter from said global
± 13 €1			communication network at a ground based station; and
9 CO 9 CO 14	,		archiving said aircraft performance parameter at said ground based station.
14			7
1	>	19.	A method for determining whether to issue an aircraft maintenance
2			advisory according to claim 18 including the steps of:
3			performing the method of claim 18;
4			analyzing said performance parameter;
5			transmitting an aircraft maintenance advisory when the analysis of said
6			performance parameter indicates an aircraft problem;

receiving said maintenance advisory on said aircraft; and

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1	340/945t	20.	An in-	flight advisory system comprising:
2	I		a trans	smitter for transmitting an advisory to an aircraft in a digital form;
3			a rece	iver located in said aircraft configured to receive said digital form of
4 <mark>1 4</mark>				said advisory; and
50			a disp	lay means for displaying said advisory in said aircraft,
4000 504 640 504			where	in said advisory includes an identifier exclusive to said aircraft.
1		21.	The i	n-flight advisory system of claim 20 wherein said advisory
2 []	comp	rises inf	ormatio	on selected from the group consisting of:
3 14			(a)	weather information;
4			(h)	air traffic control information; and
5			(i)	area traffic data.
1		22.	An in-	flight advisory system comprising:
2			a trans	smitter for transmitting an advisory to an aircraft in a digital form;
3			a rece	iver located in said aircraft configured to receive said digital form of

said advisory; and

displaying said maintenance advisory on said aircraft.

5		a disp	lay means for displaying said advisory in said aircraft,
6		where	in said advisory comprises information selected from the group
7			consisting of:
8		(a)	flight separation information;
9		(b)	topographical information;
10 <del>  -</del>		(c)	wind shear information;
10 5 11 5 12 5 12 10 13 0	ht -	(d)	lightning information;
12 =	exist.	(e)	emergency information;
13 🗓		(f)	crash avoidance information;
14 L 15 L 15 L 16 L	:	(g)	information from the manufacturer of said aircraft;
15 [1]		(h)	air traffic information;
16 <b>↓</b>		(i)	area traffic information;
17		(j)	safe to take off information; and
18	1	(k)	safe to fly information.
1	23.	An in-	flight advisory system comprising:
2		a trans	smitter for transmitting an advisory via a global communication

network;

a receiver located in said aircraft configured to receive said advisory; and

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A digital data communication system for an aircraft comprising:

a display means for displaying said advisory in said aircraft.

a transceiver located on the aircraft, said transceiver configured to transmit and receive digital data to and from a global communication network; and

a central station configured to transmit and receive digital data to and from said global communication network,

wherein a transmission by an aircraft on said global communication network includes an identifier, said identifier being unique to a particular aircraft.

25. The digital data communication system of claim 24 further comprising: a sensor multiplexer having a plurality of inputs for receiving information from a plurality of aircraft sensors, said information including the position and heading of said aircraft, and an output for digitally communicating said information to said transceiver for transmission via said global communication network.

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PATENT Attorney Docket No.: 57127 Preliminary Amendment Page 10 of 23

26.	The digital data communication system of claim 25 wherein said plurality
of aircraft sen	sors includes a GPS receiver.

- 27. The digital data communication system of claim 25 wherein said plurality of aircraft sensors includes an acoustic sensor for receiving audible information.
- 28. The digital data communication system of claim 24 further comprising a display means on said aircraft, said display means configured to display information encoded in said digital data received by said transceiver.
- 29. The digital data communication system of claim 24 wherein said central station includes data storage and at least a portion of said digital data transmitted from said aircraft is stored in said data storage.
- 30. The digital data communication system of claim 29 wherein said portion of said digital data includes data selected from the group consisting of:

airspeed of the aircraft;

- (b) aircraft attitude;
- (c) fuel status of the aircraft;

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PATENT Attorney Docket No.: 57127 Preliminary Amendment Page 11 of 23

(d) engine status of the aircraft;

(e) flight control positions;

(f) landing gear status; and

(g) control surface positions.

- 31. The digital data communication system of claim 30 wherein said portion of said digital data is analyzed at said central station to determine if a flight safety advisory or a maintenance advisory is warranted.
- 32. The digital data communication system of claim 31 further comprising a display means on said aircraft, wherein said central station transmits said flight safety advisory or said maintenance advisory to said transceiver and said display means is configured to display said flight safety advisory or said maintenance advisory.
- 33. A digital data communication system for an aircraft comprising a receiver configured to receive a transmission from a central station, said transmission being relayed to said receiver by way of a satellite and said transmission comprising digitally encoded information, wherein said digitally encoded information includes an identifier unique to a particular aircraft.

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The digital data communication system of claim 33 wherein said digitally 34. encoded information includes weather information.

35. The digital data communication system of claim 33 wherein said digitally encoded information includes maintenance advisory information.

36.

a central ground based station having a data storage device, wherein said sensor multiplexer receiver and transmitter receives aircraft performance and control parameters from existing sensors on an

A telemetric crash data recorder comprising:

a sensor multiplexer receiver and transmitter; and

aircraft and transmits said performance and control parameters to

said central ground based station over a world wide

communication system for storage in said data storage device.

- 37. The telemetric crash data recorder of claim 36 further comprising:
  - a GPS receiver in communication with said sensor multiplexer receiver
    - and transmitter such that a position of said aircraft is transmitted to

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1		38.	The telemetric crash data recorder of claim 37 wherein said central ground
2			udes a processor for analyzing performance and control parameters and said
3	n. 2m ) Dert	aircraft pos	sition such that, in the event of a crash, said processor will calculate a crash
4 <sup>‡</sup>	ν. 	site.	
1 =		39.	The telemetric crash data recorder of claim 37 wherein said performance

said central ground based station.

The telemetric crash data recorder of claim 37 wherein said performance 39. and control parameters comprise information recorded by an on board flight data recorder.

a radio frequency transceiver located on an aircraft, said radio frequency transceiver configured to transmit and receive digital information; an inertial navigation system located on said aircraft, said inertial navigation system providing the position of said aircraft to said

An air traffic control system comprising:

transceiver; and an air traffic control facility configured to receive and display said position of said aircraft to an air traffic controller.

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PATENT Attorney Docket No.: 57127 Preliminary Amendment Page 14 of 23

	41.	The air traffic control system of claim 40 further comprising a GPS
receiv	er, when	ein said position of said aircraft is augmented with data from said GPS
receiv	er.	

42. An improved air traffic control system of the type having an air traffic control facility wherein air traffic controllers observe a radar image of controlled aircraft, the improvement comprising:

a ground based station comprising:

a receiver for receiving precision navigation information from the controlled aircraft;

a ground communication system in communication with the air traffic control facility,

wherein said precision navigation information is transmitted to the air traffic control facility over said ground communication system to enhance the information provided to the air traffic controllers.

43. The improved air traffic control system of claim 42 wherein said ground communication system includes a fiber optic link between said ground based station and

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a positioning system for providing an aircraft position;

a transceiver for transmitting said aircraft position and receiving a position of nearby aircraft;

a display for displaying said position of nearby aircraft relative to said aircraft position,

A ground collision avoidance system for an aircraft comprising:

wherein each aircraft which transmits an aircraft position has a unique identifier and said unique identifier is included in each transmission.

- 45. The ground collision avoidance system of claim 44 wherein said positioning system comprises an inertial navigation system.
- 46. The ground collision avoidance system of claim 45 wherein said positioning system further comprises a GPS receiver.

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47.	The ground collision avoidance system of claim 45 wherein said aircraft
includes a flig	tht data recorder and said positioning system provides data to said flight
data recorder.	

48. The ground collision avoidance system of claim 45 wherein said positioning system further provides an aircraft heading and wherein said transceiver transmits said aircraft heading.

A method for transmitting and receiving aircraft performance and control 7 7013 49. parameters comprising:

> providing a sensor multiplexer on an aircraft for receiving information from aircraft sensors;

transmitting said information to a communication network with a first transceiver aboard said aircraft, said first transceiver configured to transmit digital information on said communication network; and receiving said digital information at a ground station having a second transceiver configured to receive information from said communication network.

	1	101/3	50.	An aircraft having a global digital communication system comprising:
	2	·		a transceiver for digital communication over a global communication
	3			system;
	4			an in-cockpit display having a display means, said display means receiving
	5 <mark>  4</mark>			flight advisory data from said transceiver, and an operator input
	О 6 <u>С</u>			means; and
•	50004400 7400 8 4000 1000 1100			a multiplexer for receiving information from aircraft sensors and from said
)	8 🗓			in-cockpit display, said multiplexer having an output in
	95			communication with said transceiver for transmitting said
	10 M			information over said global communication network,
	IJ 11 <b>}</b> ±			wherein said information comprises aircraft performance and control
	12			parameters provided to a flight data recorder on board the aircraft.
	1		51.	The aircraft of claim 50 wherein said flight advisory data includes at least

one advisory from the group consisting of:

(a)

(b)

(c)

weather advisory;

air traffic advisory;

anti-collision advisory; and

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- (d) ground incursion advisory;
- (e) flight information advisory;
- 52. The aircraft of claim 50 wherein said flight advisory information includes a maintenance advisory wherein said maintenance advisory is transmitted from a ground station to said transceiver over said communication upon the receipt and analysis of said information.
- 731/301 53. A method for avoiding an in-flight collision including the steps of:
  - (a) transmitting an identifier from an aircraft, said identifier being unique to said aircraft;
  - (b) transmitting the position of said aircraft wherein said position of said aircraft includes a heading of said aircraft;
  - (c) receiving said identifier and said position at a central station;
  - (d) analyzing said position of said aircraft relative to other objects and the ground to determine a risk of collision;
  - (e) sending a flight safety advisory to said aircraft when said analysis indicates there is a risk of collision.

PATENT Attorney Docket No.: 57127 Preliminary Amendment Page 19 of 23

1	54.	The method of claim 53 wherein the position of step (b) further includes:
2		the altitude of said aircraft;
3		the latitude of said aircraft;
4		the longitude of said aircraft; and
1 5	55.	The method of claim 54 wherein the position of step (b) further includes:
20		the pitch position of said aircraft; and
1 D 2 D 3 P 1 D 1 D 2 D 3 P 3 P		the roll position of said aircraft.
70		
1 4	56.	The method of claim 55 wherein the position of step (b) further includes;
2 [[]		the rate of climb of said aircraft;
3 <u>L</u>		the velocity of said aircraft;
4		the yaw rate of said aircraft;
5		the pitch rate of said aircraft; and
6		the roll rate of said aircraft.
1	57.	The method of claim 53 wherein step (b) includes the substeps of:
2		(b)(i) obtaining a position of said aircraft from the inertial reference
3		system of said aircraft; and

(b)(ii) transmitting said position of said aircraft wherein said position of

said aircraft includes a heading of said aircraft.

The method of claim 57 wherein step (b) includes the substeps of:

(b)(ii) obtaining a position of the flight controls of said aircraft;

(b)(i) obtaining a position of said aircraft from the inertial reference

(b)(iii) transmitting said position of said aircraft and said position of flight

controls of said aircraft wherein said position of said aircraft

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includes a heading of said aircraft.

59. The method of claim 53 including the additional steps of:

(f) receiving said flight safety advisory on said aircraft; and

(g) displaying said flight safety advisory to the flight crew of said aircraft.

60. The method of claim 53 wherein step (d) includes the substeps of:

(d)(i) calculating a separation distance between said aircraft and a

plurality of other objects; and

system of said aircraft;

PATENT Attorney Docket No.: 57127 Preliminary Amendment Page 21 of 23

4		(d)(ii)	analyzing said separation distance and the position of said aircraft
5			relative to the ground to determine a risk of collision.
1	61.	The m	ethod of claim 60 further including the steps of:
2		(f)	sending said separation distance to said aircraft;
3 <b>1</b> 4 <b>1</b>		(g)	displaying said separation distance to the flight crew of said
	aircraft.		
1 <del>  1</del>	340/945 62.	A safe	to take off advisory system comprising:
1 TO CONTRACTOR OF THE CONTRAC		a trans	ceiver located in said aircraft configured to transmit aircraft
<b>₽</b> 3  4			performance and control parameters and to receive a safe to take
, <del>1</del>			off advisory; and
5		a centi	ral station for receiving said aircraft performance and control
5			parameters and transmitting said safe to take off advisory to an
7			aircraft based on said performance and control parameters;
3		a displ	ay means for displaying said safe to take off advisory in said
)			aircraft, .